

January 18, 1930

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AVIATION

The Oldest American Aeronautical Magazine

Regulating AMERICA'S AIR COMMERCE

FORECASTING *Aircraft Demand*

AVIATION PROGRESS IN *Soviet Russia*



IN TWO SECTIONS
Section One

[illegible]

From Kansas City
One of America's great air-centers
go 46 routes

Year	Age	Sex	Height (cm)	Weight (kg)	Body Mass Index (kg/m ²)	Waist Circumference (cm)	Hip Circumference (cm)	Waist-Hip Ratio	Trunk Fat (%)	Visceral Fat (cm)	Subcutaneous Fat (cm)	Visceral Fat Index (cm ³)	Subcutaneous Fat Index (cm ³)	Visceral Fat to Subcutaneous Fat Ratio
1992	20	M	170	65	22.0	85	95	0.89	15	10	15	100	100	1.0
1993	21	M	172	68	22.5	88	98	0.90	16	11	16	110	110	1.0
1994	22	M	174	70	22.7	90	100	0.90	17	12	17	120	120	1.0
1995	23	M	176	72	22.9	92	102	0.90	18	13	18	130	130	1.0
1996	24	M	178	75	23.0	95	105	0.90	19	14	19	140	140	1.0
1997	25	M	180	78	23.3	98	108	0.90	20	15	20	150	150	1.0
1998	26	M	182	80	23.6	100	110	0.91	21	16	21	160	160	1.0
1999	27	M	184	82	23.9	102	112	0.91	22	17	22	170	170	1.0
2000	28	M	186	85	24.2	105	115	0.91	23	18	23	180	180	1.0
2001	29	M	188	88	24.5	108	118	0.91	24	19	24	190	190	1.0
2002	30	M	190	90	24.7	110	120	0.92	25	20	25	200	200	1.0
2003	31	M	192	92	25.0	112	122	0.92	26	21	26	210	210	1.0
2004	32	M	194	95	25.3	115	125	0.92	27	22	27	220	220	1.0
2005	33	M	196	98	25.6	118	128	0.92	28	23	28	230	230	1.0
2006	34	M	198	100	25.8	120	130	0.93	29	24	29	240	240	1.0
2007	35	M	200	102	25.5	122	132	0.93	30	25	30	250	250	1.0
2008	36	M	202	105	26.0	125	135	0.93	31	26	31	260	260	1.0
2009	37	M	204	108	26.3	128	138	0.93	32	27	32	270	270	1.0
2010	38	M	206	110	26.5	130	140	0.93	33	28	33	280	280	1.0
2011	39	M	208	112	26.5	132	142	0.93	34	29	34	290	290	1.0
2012	40	M	210	115	26.2	135	145	0.93	35	30	35	300	300	1.0
2013	41	M	212	118	26.3	138	148	0.93	36	31	36	310	310	1.0
2014	42	M	214	120	26.2	140	150	0.93	37	32	37	320	320	1.0
2015	43	M	216	122	26.0	142	152	0.93	38	33	38	330	330	1.0
2016	44	M	218	125	25.7	145	155	0.93	39	34	39	340	340	1.0
2017	45	M	220	128	25.5	148	158	0.93	40	35	40	350	350	1.0
2018	46	M	222	130	25.7	150	160	0.94	41	36	41	360	360	1.0
2019	47	M	224	132	25.8	152	162	0.94	42	37	42	370	370	1.0
2020	48	M	226	135										



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A DAY to the west lies Los Angeles. A half day to the east—Cleveland. To Dallas in 7 hours. To Chicago in 4. Here in Kansas City, where 45 routes now converge, there has developed one of the great centers of America's new transport system—the air. A glance at the combined timetables of these transport lines reveals a significant

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Wasp & Hornet & ENGINES

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MANUFACTURERS' SPECIFICATIONS ON AMERICAN C

THE TABLE BELOW IS BELIEVED TO BE ACCURATE BUT A VARIOUS

This table will appear monthly and :

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COMMERCIAL AIRPLANES AS COMPILED BY AVIATION

FA DOES NOT ASSUME RESPONSIBILITY FOR THE FIGURES GIVEN

overlaid and equalized are shown

Performance with Full Load at Sea:
Based on Standard 100 Measurements

1	2	3	4	5
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100

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1. **Introduction**
 2. **Background**
 3. **Methodology**
 4. **Results**
 5. **Conclusion**
 6. **References**

100

1. **Introduction**
 2. **Background**
 3. **Methodology**
 4. **Results**
 5. **Conclusion**
 6. **References**

THE
NEW
WORLD

100



1998

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Figure 1

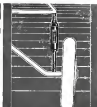
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Figure 1 consists of two main panels, (a) and (b), each containing three bar charts. Panel (a) shows the probability of being employed for men, and panel (b) shows it for women. Each panel has three charts for different age groups: 18-24, 25-34, and 35-44. The x-axis for all charts is 'Year' with categories 'Before 2008', '2008', and 'After 2008'. The y-axis is 'Probability of being employed' ranging from 0 to 1.0. In panel (a), the probability for men generally decreases after 2008, especially for the 18-24 age group. In panel (b), the probability for women also decreases after 2008, but the decline is less pronounced than for men, particularly for the 18-24 age group.

[illegible]

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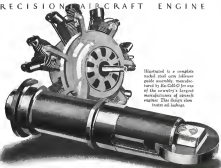
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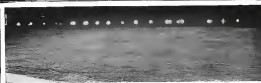
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SIKORSKY AMPHIBION



AVIATION

THE OLDEST AMERICAN AERONAUTICAL MAGAZINE

A McGRATH-PALL PUBLICATION • ESTABLISHED 1918

EDWARD P. WARNER, Editor

Volume 10 • January 15, 1930 • • • Number 1



A Word for the Forgotten Man

BEFORE the most significant feature of the army surplus passed to our recent inquiry on the depletion of the ranks of government technical aviation personnel was the absolute futility with which the commercial industry approached the problem. In fact, the straight thinking and general absence of self-interest was so refreshing contrast to the manner in which government departments have in some cases ostentatiously handled the matter.

The obvious remedy, the raising of government salaries to place them on a comparable basis with commercial compensation for equivalent work, has often been mentioned. Turned to an extreme, this involves the need of a basic change in civil service classifications or of special legislation favorable to aviation personnel alone. Yet, the real crux of the problem is a matter of cases in the government aviation services is not to secure pay higher than other equivalent branches of government service, but only to assure the payment of rates already established. With this only accomplished, commercial competition for subordinate personnel is not commonly so met on a comparable footing.

In order to establish the setting for an analysis indicating how the government's interests may be protected without the need for heroic action, it should be noted that while the civil service commission establishes the qualifications for a given class of employment the rates of pay are set by a different body. In many cases where there are one or more rates of pay tabulated for a given title or classification, the upper rates are withheld from aviation personnel, and in most instances a majority of those in a classification are actually paid rates falling within the lower third of the available range. The press has at times heralded alleged increases in established legal rates of pay when in reality the fact was rates were merely interpolated between existing rates. The practical result

was that when a man was recommended for promotion it would take two separate raises to achieve what formerly had been accomplished by one. The authority which establishes legal rates of pay is entirely divorced from the authority that applies them, and the higher rates may exist for years on the books without anyone actually being paid them.

For years, despite a wide range of rates of pay legally established for each classification of duty, raises have been made almost solely when resignations were effected. Furthermore, while a new man quite unknown to the local activity might be appointed direct from a civil service registry to fill a vacancy high up in the organization, a man already in the employ of the local activity, who had similarly qualified by passing the civil service examination for the higher rank, required reference to headquarters, often involving months of time and volumes of correspondence to effect the raise.

Now when resignations came so fast as they did during the past year, this machinery of promotion became entirely too involved, and while quotas were kept up approximately by increased local activity, the average rate of pay of the personnel rapidly slumped until it was and is far below that of other government agencies similarly functioning. And the situation has been further complicated by a general order, now being reviewed, that specific raises are only to be considered once a year, and that in cases of resignations only one promotion is to be effected, instead of a general moving up all along the line.

While such a ruling may be entirely fair in long-established and strictly military activities, or activities related to massive industries such as ship-building, where personnel turnover is at a minimum, it works to the distinct disadvantage of the aviation civil service personnel. Their rates of pay now average, and are

likely to confuse is average, far less than the other activities, thanks to the far greater aerial maneuver of personnel in a service with an extremely active commercial counterpart.

Such at least is the situation in one leading governmental aviation activity. Possible steps to meet the situation and forward further dignities of personnel are almost obvious. Less red tape and the placing of more authority for apportioning expenditures for personnel in the hands of those directly charged with aviation developments, subject of course to the limitations of budget expenditures and civil service regulations for qualifications, would do much to alleviate the situation. If government aviation activities really paid can be brought on a parity with those of other government activities, most of the differential which now attracts personnel to commercial industry will be automatically removed. There is serious matter here for the consideration of wage boards, local civil service boards, and the heads of the Departments immediately concerned—and the manner of the aircraft industry is obvious.

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Collisions in Mid-air

THE recent mid-air collision of two twin planes engaged in morning practice work near Los Angeles, with the resultant loss of ten lives, has horrified the country. Another collision a year ago, in which a transport machine was involved and in which there was again heavy loss of life, had an equally shocking effect. The saddest of collisions, when descending safety may come without an instant's warning and under the best of flying conditions, takes it more than seriously threatening. Although such accidents are comparatively rare, they cannot by any means be disregarded by those responsible either for the planning of aircraft operation or for the development of airplane design.

Two planes were flying parallel about five hundred feet apart when one, slightly above and ahead of the other, was thrown into a bank and turned into its compass as an angle of approximately forty-five degrees. It appears from somewhat meager accounts that the planes met at an angle completely blind for one and partially blind for the other, and thereby lost the lesson.

The open cockpit, which for a long time was the only standard type, had a substantial blind area forward and below the lower wing. Everyone knew it. It advertised itself, and thence by its safety. A pilot was ever conscious of the blind area, and he was on the lookout.

No longer does the open-cockpit airplane hold the field alone. New types have come into use, and have increasingly improved vision, on a certain score,—on one of their major merits. Planes affording perfect clarity of vision along the flight path and for a considerable dis-

tance on every side of it have become commonplace, but to every improvement brings a measure of it leads us to assume that the danger of collision has been eliminated.

There are a number of types of planes in the air, flying qualities, and with excellent vision to the limit, which are almost completely blind when turning. As soon as the plane is banked, the wing cuts off the pilot from all view of the region into which he is swinging, and apparently that has behind the recent disaster. No one will ever know as to its details exactly what happened over the fringe of the Pacific, but observation of many cases in which an accident did result leaves no doubt that many pilots otherwise skilled and carefully trained had been faced with the effective vision characteristics of a plane as modified by banking and changing course. The writer of the editorial has frequently been astonished, and when he was riding as a passenger more than a mile aloft, at the amazement with which a pilot flying over a crowded airport would go into a steep bank and a sharp turn with a plane in which he not below the wing, and without any preliminary survey of the region into which he was about to turn.

There is no occasion for collisions. The first requirement is care, care, and yet more care upon the part of every pilot. The air traffic rules must be obeyed, and that they are not obeyed by individual individuals they must be enforced. Special provisions must cover airports at which traffic is unusually dense, and they too must have unquestioning and inevitable compliance.

Assuming that the pilot will do his best, there remains a responsibility for the designer. With all due credit to the strides that the aeronautical art has made, it must be admitted that vision is still generally unsatisfactory. Designers and designers must give more attention to vision of ground commercial flying is to become popular. Vision from an airplane in flight is not a luxury or a secondary consideration. It is of prime importance and the basis of all safe operation, for unless we can see our dangers we certainly cannot be expected to avoid them. Vision should be greatly improved, and must be, even at the sacrifice of other factors, if progress is to continue.

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Common Sense and Selling

THERE HAS BEEN much loose talk in the aviation industry about the selling of airplanes, with the general theme that all the industry needs to do is to develop an "effort" sales organization and "infiltrate the market" in order to dispose of all the airplanes that can be produced. It is hardly necessary to point out that the electric refrigerator industry might devote a vast amount of money and effort to the cultivation of sales in Greenland without external result.

Certainly in selling airplanes we might as well admit

at the start that since airplanes cannot be sold to anyone and that some people will not buy any kind of an airplane, regardless of price or merit. Assuming as extreme of pessimism, it is conceivable that there might be an immediate market of any material importance for any sort of an airplane. As a matter of fact it must be admitted that the market is very definitely bounded as to types of planes and as to groups or social classes from which the purchasers are to be drawn. While we may not be able to predict how many planes or exactly what types of planes these groups will purchase, we can definitely rule out as the most extreme outside of our groups of possible purchasers, and every plane not suited for sale to one of these groups, so far as commercial production and sale are concerned.

Without attempting any extensive analysis, it is almost self-evident that there are four main groups purchasing aircraft. They are: individuals financially able to buy for sport or personal travel, individual business firms buying planes for their own use, transport lines, flying schools, or miscellaneous "aerial service" operators. The only planes which can be produced commercially are those which fit the requirements of buyers contained in the four groups given above. Consequently, all sales efforts should be directed to those groups, and allocated among them with some discrimination. Any effort to sell planes to commercial quantities to the private owner must still be speculative, a planning of need for later harvest. A new habit will have to be developed in a large section of the community,—which takes time. Few individual builders of aircraft, however large and ambitious, and patriotic they may be, can afford to make everything in the immediate future upon the success of a one-man campaign for the private purchase of aircraft. "Democratization" is a slogan the value of which is not limited to the business wheat farmers.

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Private Flying Clubs

FOR SEVERAL YEARS now Americans have been observing with admiration and envy the progress made in various parts of the British Empire for providing private flying sport upon the most reasonable financial terms. The light plane clubs, fostered by the government of Great Britain, and subsequently by most of the British Dominions, have provided for the activities of some thousands of pilots who would have had no opportunity at all of flying on other terms.

The British Empire has no patent on this scheme. Although there is not the remotest prospect of duplicating in the United States the subsidies which the British and Dominion governments have provided, those have not been large enough to be necessary devices. The basic principle should work as well upon America soil as it

Canada or Australia. The plan of divided ownership of an airplane by a group of people who are financially unable at least for some other reason unwilling to become involved ought to have taken hold in America before now.

For some reason it has not done so. There is a great deal of talk about flying clubs, but comparatively few of them are actually flying. The plan of the National Aeronautic Association and that of the Aeronautic Club of Ohio, the two most prominently organized on a national scale, have both been very slow to carry out their scope. Individual clubs detached from any national organization have in most cases, although not in all, used a method of management and control of an inadequate financial protection against mishap, and many have been wrecked in the early stages of their voyage.

For the private flying club enthusiast who looked only backward the prospect would be more than gloomy, but we do not despair of the future. We believe that the club idea is fundamentally sound, at least as a means of helping private flying to a sport through its early stages. We believe that this country can support at least five hundred constantly active and self-sustaining clubs. We hope to see them formed and operated under proper conditions.

As in all other recreational activities, it is folly to rush ahead without recognition of facts. Clubs cannot be organized on a wing. They are not going to run themselves. If they attempt to get along without insurance, they are taking a chance on group of business men ought to assume, and they are making the needs of recreation and interest dependent. The greatest obstacle to the formation of flying clubs has been the individual, self-defined or led army by professional agitators, who signify fitness that it ought to be possible to fly for an average pilot cost of about \$2 an hour and throw all possibility of accident out of his reckoning.

There are a great number of would-be pilots who have no desire to go into flying as a profession, and who at present achieve little taking instruction because they see nothing of interest at the end of their school course. There are many already qualified who have reluctantly abandoned their aerial projects. They need the flying club, and we know no other device that can serve their interest in the same fashion. The industry needs the flying club as an means of promoting the use of its products. It needs it desperately. Manufacturers of aircraft, airlines, operators of airfields, and all those whose future depends upon the increasing of public interest in personal use of the airplane should do everything in their power to put an added drive behind the organization of clubs on a national scale, and under enough central control to assure some safety and unity in their operation. The National Aeronautic Association and the Aeronautic Country Clubs, as well as any other organization which may usefully undertake to promote club flying upon reasonably intelligent lines, deserve not only the aircraft industry's sympathy but its whole-hearted and active support.

REGULATING Air Commerce



By J. S. MAIRRUTT

Chief, Inspection Service, Aeronautics Branch,
Department of Commerce

ARTICLE I—

INSPECTION

divided into nine districts with a supervising aeronautical inspector in charge of each district. There is one office centrally located in the area for which each supervisor is responsible. Fig. 1, shows the division of the United States into the nine inspection districts.

The names of the nine supervisors, with the addresses of their district offices, are as follows: G. R. Goodson, Dept. of Commerce, Roosevelt Field, Garden City, L. I., N. Y.; C. A. Charles, Central Airport, Camden, N. J.; Leo C. Wilson, Municipal Airport, Athens, Ga.; R. D. Brodages, 602 First Press Building, Detroit, Mich.; C. W. Vent, Municipal Airport, 9332 South Cicero Ave., Chicago, Ill.; R. H. Lees, Jr., Room B, Chamber of Commerce, Kansas City, Mo.; H. B. Postland, Army Airfield, Love Field, Dallas, Tex.; E. E. Moulton, Oakland Airport, Oakland, Calif.; W. F. Purdon, 629 Chamber of Commerce Building, Los Angeles, Calif.

Besides the Washington office personnel and the district supervisors, there are 38 aeronautical inspectors, 8 aeronautical engineering inspectors, 4 aeronautical school inspectors, and 16 (factory) airplane inspectors. With the exception of the (factory) airplane inspectors, all of the above inspection personnel are pilots with years of experience. Men who on account of the nature of their work must not only possess a thorough knowledge of airplanes and airplane construction and

show the average ability as a pilot, but also sufficient skill and diplomacy to properly meet the public and carry out the work of examining pilots and mechanics, and of inspecting aircraft in the field. The airplane inspectors stationed at the various aircraft factories are selected for their intimate knowledge of structural details in the manufacture of aircraft and are not necessarily pilots. Over 80 per cent of the aircraft production is now



The boundaries of Commerce airplane inspection districts of the United States

covered by (factory) airplane inspectors, which means that all such aircraft leave the factory already inspected, licensed and ready for commercial operation.

THE AERONAUTICAL INSPECTORS examine pilots and mechanics of license. They investigate accidents, examine witnesses in connection therewith, and file complete reports giving as accurately as possible the cause of the accident. They report violations of the Regulations and Air Traffic Rules and recommend the penalty to be assessed, they are even authorized to lift the license of a pilot or pilot if the violation is serious enough to warrant such action.

The engineering inspectors inspect and test in actual flight all new types of aircraft to determine their eligibility for an approved type certificate, or for license. These men of necessity must be exceptional pilots and in addition possess a knowledge of aeronautical engineering. There is added a routine phase of the work, putting new models through their paces under all loading conditions, testing them for performance, balance, stability, maneuverability, and recovery from spin. On several occasions, history test pilots working with the engineering inspectors on the tests of some new or radical type have had to leave the plane by parachute in order to save their lives. In addition to testing the plane itself, the engineering inspectors inspect the factory for facilities, workmanship and material.

Recently, the Department provided in the Regulations for the approval of airplane repair stations. The inspection of these aircraft repair and service stations will also be done by the engineering inspection personnel.

Aeronautical school inspectors are responsible for the examination of the instruction personnel of the aviation school, both the flying instructors and the ground instructors, as well as for the inspection of the equipment and facilities of the school itself. The school inspection work is relatively new. The standard to the Air Commerce Act of 1930 authorizing the inspection of flying schools when such schools apply was only passed Feb. 28, 1929. However, quite reasonable and thoroughly gratifying results have been accomplished since that time. At the time of writing, 24 schools have been listed approved



Fig. 1. Organizational Chart of the regulatory activities of the Aeronautics Branch. These activities, together with Airframe and Aeronautical Engineering, are under the supervision of the National Secretary of Commerce for Aeronautics.

The efficient and satisfactory regulation of air commerce is a gigantic task which has been handled by the Aeronautics Branch since the passing of the Air Commerce Act in 1926. Although handicapped by limited appropriations and a shortage of personnel, the Branch has expanded rapidly to keep pace with the growing industry. AVIATION welcomes this opportunity to present in six articles the activities in connection with the regulatory work

of the Branch, headed by Gilbert G. Bixding as director of Air Regulations. They come with an irrefragable authority over the signatures of the chiefs of the several branches concerned. The first article deals with that all important item of inspection. The second article which will appear in next week's issue, has been written by Kenneth M. Lane, who tells about the organization and work of the Engineering Section of which he is chief.

Department of Commerce, with the exception of the Civil sub-industry from 1925 to June 30, 1929, illustrates the shape division of aircraft distribution. The significant change from 80 per cent of the total annual output going to the Government in 1920 to approximately 80 per cent of total production now going into the commercial field is evidence of a wider adaptation of aviation

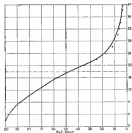


Figure 1—Percentage of plane production by type. The percentage of plane production by type which will remain in use in succeeding periods.

on the part of the general economy. These distributions are based on total numbers of planes. Owing to the higher unit cost of military planes, production for military use is still about half the aggregate value.

Before taking up the factor of replacement a glance

at Charts I and II are reproduced, vividly displays this fact of concentration of demand underlying the industry. Fully 62 per cent of which, it appears, can be confidently looked upon as stable continuous annual demand. The evidence upon which these references rest is to be found through an analysis of the accompanying charts and tables, which represent, so far as the writer knows, the most authentic records available.

The making of a careful analysis of distribution requires a full appreciation of the general nature and individual character of the market. Such an analysis should be directed always at both sides of the market, the supply side and the demand side. However, in this study we are first concerned with trying to arrive at a general impression of the market's underlying character and must leave its more detailed analysis to a more comprehensive study later.

From the standpoint of a wider public usage of aircraft services the demand of last has been the significant increase in the use of the aircraft, especially after the advent of the private operating company. The increase in this division has been 3,000 per cent during the period from January, 1928, to the end of August, 1929. In other words the community is now forwarding via the air just thirty times more mail per month than they were an equal number of months back. The increase in express and freight shipped by air has risen from a few thousand pounds in 1924 to a total monthly sum in excess of 300,000 lb., according to official Department of Commerce reports. Also in number of passengers carried, exclusive of field or airport society hops, the increase has been even more substantial.

In order that we may picture the full extent of the total use of aircraft for commercial purposes today we can best arrive at an estimate by adding the three divisions of mail, express and passengers in terms of pounds carried. For these combined, the total monthly volume approximates 1,500,000 lb., or nearly 1,000 tons of paid transport service through the air monthly. The total for 1929 will probably greatly exceed 24,000,000 lb.

Table I—Comparative Annual Aircraft Production and Consumption Totals											
Total Annual Aircraft Production						Annual Aircraft Consumption					
Total production does not include planes, engines or parts for replacement of and in each year's year						Total consumption does not include planes, engines or parts for replacement of and in each year's year					
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the resulting cost of the inefficient producer, learned will continue to ride its tail. The larger concern will have to show up as they reduce their bids for labor against the pressure of the industry. Likewise, large-scale consolidation will result in stabilization of production and prices, and tend to smooth out the peaks and valleys of output, which must inevitably grow highly beneficial.

Notes that the structure of the industry is becoming more or less completely monopolized, in every branch, on basically sound business lines it seems destined to achieve a success of considerable magnitude. At least that much is the expectation. This fusion of capital, methods, ideas, designs, new equipment means combined with the welding together of parts manufacturing units, such as producers of carburetors, magnets, etc., which enter copiously to the automobile as well as the aeronautical trades, assures quality production advantages, large-

Table 10.—Percentage Analysis of Changes in Production, 1925-1935

Year	Total Shipment	Out of Commercial										Total Out- er Cost
		Government		Export	Transit		Private		Domestic	Foreign	Total	
		For Cost	For Cost		For Cost	For Cost	For Cost	For Cost				
1925	128	40	11	11	10	10	10	10	10	10	10	10
1926	130	40	11	11	10	10	10	10	10	10	10	10
1927	130	40	11	11	10	10	10	10	10	10	10	10
1928	130	40	11	11	10	10	10	10	10	10	10	10
1929	130	40	11	11	10	10	10	10	10	10	10	10
1930	130	40	11	11	10	10	10	10	10	10	10	10
1931	130	40	11	11	10	10	10	10	10	10	10	10
1932	130	40	11	11	10	10	10	10	10	10	10	10
1933	130	40	11	11	10	10	10	10	10	10	10	10
1934	130	40	11	11	10	10	10	10	10	10	10	10
1935	130	40	11	11	10	10	10	10	10	10	10	10
1936	130	40	11	11	10	10	10	10	10	10	10	10
1937	130	40	11	11	10	10	10	10	10	10	10	10
1938	130	40	11	11	10	10	10	10	10	10	10	10
1939	130	40	11	11	10	10	10	10	10	10	10	10
1940	130	40	11	11	10	10	10	10	10	10	10	10
1941	130	40	11	11	10	10	10	10	10	10	10	10
1942	130	40	11	11	10	10	10	10	10	10	10	10
1943	130	40	11	11	10	10	10	10	10	10	10	10
1944	130	40	11	11	10	10	10	10	10	10	10	10
1945	130	40	11	11	10	10	10	10	10	10	10	10
1946	130	40	11	11	10	10	10	10	10	10	10	10
1947	130	40	11	11	10	10	10	10	10	10	10	10
1948	130	40	11	11	10	10	10	10	10	10	10	10
1949	130	40	11	11	10	10	10	10	10	10	10	10
1950	130	40	11	11	10	10	10	10	10	10	10	10
1951	130	40	11	11	10	10	10	10	10	10	10	10
1952	130	40	11	11	10	10	10	10	10	10	10	10
1953	130	40	11	11	10	10	10	10	10	10	10	10
1954	130	40	11	11	10	10	10	10	10	10	10	10
1955	130	40	11	11	10	10	10	10	10	10	10	10
1956	130	40	11	11	10	10	10	10	10	10	10	10
1957	130	40	11	11	10	10	10	10	10	10	10	10
1958	130	40	11	11	10	10	10	10	10	10	10	10
1959	130	40	11	11	10	10	10	10	10	10	10	10
1960	130	40	11	11	10	10	10	10	10	10	10	10
1961	130	40	11	11	10	10	10	10	10	10	10	10
1962	130	40	11	11	10	10	10	10	10	10	10	10
1963	130	40	11	11	10	10	10	10	10	10	10	10
1964	130	40	11	11	10	10	10	10	10	10	10	10
1965	130	40	11	11	10	10	10	10	10	10	10	10
1966	130	40	11	11	10	10	10	10	10	10	10	10
1967	130	40	11	11	10	10	10	10	10	10	10	10
1968	130	40	11	11	10	10	10	10	10	10	10	10
1969	130	40	11	11	10	10	10	10	10	10	10	10
1970	130	40	11	11	10	10	10	10	10	10	10	10
1971	130	40	11	11	10	10	10	10	10	10	10	10
1972	130	40	11	11	10	10	10	10	10	10	10	10
1973	130	40	11	11	10	10	10	10	10	10	10	10
1974	130	40	11	11	10	10	10	10	10	10	10	10
1975	130	40	11	11	10	10	10	10	10	10	10	10
1976	130	40	11	11	10	10	10	10	10	10	10	10
1977	130	40	11	11	10	10	10	10	10	10	10	10
1978	130	40	11	11	10	10	10	10	10	10	10	10
1979	130	40	11	11	10	10	10	10	10	10	10	10
1980	130	40	11	11	10	10	10	10	10	10	10	10
1981	130	40	11	11	10	10	10	10	10	10	10	10
1982	130	40	11	11	10	10	10	10	10	10	10	10
1983	130	40	11	11	10	10	10	10	10	10	10	10
1984	130	40	11	11	10	10	10	10	10	10	10	10
1985	130	40	11	11	10	10	10	10	10	10	10	10
1986	130	40	11	11	10	10	10	10	10	10	10	10
1987	130	40	11	11	10	10	10	10	10	10	10	10
1988	130	40	11	11	10	10	10	10	10	10	10	10
1989	130	40	11	11	10	10	10	10	10	10	10	10
1990	130	40	11	11	10	10	10	10	10	10	10	10
1991	130	40	11	11	10	10	10	10	10	10	10	10
1992	130	40	11	11	10	10	10	10	10	10	10	10
1993	130	40	11	11	10	10	10	10	10	10	10	10
1994	130	40	11	11	10	10	10	10	10	10	10	10
1995	130	40	11	11	10	10	10	10	10	10	10	10
1996	130	40	11	11	10	10	10	10	10	10	10	10
1997	130	40	11	11	10	10	10	10	10	10	10	10
1998	130	40	11	11	10	10	10	10	10	10	10	10
1999	130	40	11	11	10	10	10	10	10	10	10	10
2000	130	40	11	11	10	10	10	10	10	10	10	10
2001	130	40	11	11	10	10	10	10	10	10	10	10
2002	130	40	11	11	10	10	10	10	10	10	10	10
2003	130	40	11	11	10	10	10	10	10	10	10	10
2004	130	40	11	11	10	10	10	10	10	10	10	10
2005	130	40	11	11	10	10	10	10	10	10	10	10
2006	130	40	11	11	10	10	10	10	10	10	10	10
2007	130	40	11	11	10	10	10	10	10	10	10	10
2008	130	40	11	11	10	10	10	10	10	10	10	10
2009	130	40	11	11	10	10	10	10	10	10	10	10
2010	130	40	11	11	10	10	10	10	10	10	10	10
2011	130	40	11	11	10	10	10	10	10	10	10	10
2012	130	40	11	11	10	10	10	10	10	10	10	10
2013	130	40	11	11	10	10	10	10	10	10	10	10
2014	130	40	11	11	10	10	10	10	10	10	10	10
2015	130	40	11	11	10	10	10	10	10	10	10	10
2016	130	40	11	11	10	10	10	10	10	10	10	10
2017	130	40	11	11	10	10	10	10	10	10	10	10
2018	130	40	11	11	10	10	10	10	10	10	10	10
2019	130	40	11	11	10	10	10	10	10	10	10	10
2020	130	40	11	11	10	10	10	10	10	10	10	10
2021	130	40	11	11	10	10	10	10	10	10	10	10
2022	130	40	11	11	10	10	10	10	10	10	10	10
2023	130	40	11	11	10	10	10	10	10	10	10	10
2024	130	40	11	11	10	10	10	10	10	10	10	10
2025	130	40	11	11	10	10	10	10	10	10	10	10

*Based on output of 1925.

scale savings in the purchase of the highest grade materials so essential to aircraft manufacture, and cost savings in operating costs, which unquestionably represents greatest forward step in the history of the industry.

Notes are made that the past three years have shown outstanding progressive and apparently persistent gains in aviation. On a broad combined output of over four thousand units in 1935 and what approximately are thousand units for 1936 the leading manufacturers will shortly report very favorable income statements despite the recent over-expansion so visible in the industry.

These are runnings within the fold to the effect that considerable doubt exists as to the probability of a continuation of the same rate of productive output that has resulted over the past few years. It can not yet be said that production is entirely regulated by demand as indicated by a close analysis of the major character of domestic consumption. It can be readily seen in Table II that during the last half of the decade there was considerable increase in the so-called "civil" or commercial demand has consumed fully 68 per cent of the total output. It is this commercial demand which just now seems to be eating most of the dividend.

Granting the truth of these assertions, there still remains a large element of doubt as to whether the essential characteristics, domestic aircraft demand may not continue to undergo marked change.

Nothing can be of more importance in clarifying this atmosphere than a true knowledge of the major elements

Table 11.—Classification of Aircraft Services

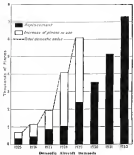
Year	Passenger		Cargo		Mail		Total	
	Number	Value	Number	Value	Number	Value	Number	Value
1925	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1926	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1927	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1928	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1929	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1930	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1931	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1932	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1933	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1934	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1935	10,170	\$1,170,000	1,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000

*Based on output of 1925.

which to the profitable business men constitutes demand. In Table I, under total consumption the classification is either broad, being divided among Government, Civil, and Export. In Table II the classification is further refined into its main constituent parts in order that the major divisions may be studied as they individually trends, their evidence of constant growth, as well as to a certain stability which may be looked to in the future. Now we shall characterize this classification and attempt to study the more fundamental elements behind the whole of the demand.

The demand for aircraft can be traced to three major sources: First, the increasing use of aviation which has its effect on the number of planes in operation; second, the elimination from use of old planes, the factor of replacement; and third, the constant replacement and engineering advances making greater safety, flexibility and utility, the factor of obsolescence.

The first factor is fairly obvious from a study of Table II, which gives the major distribution of the actual output. The ever rising demand has evidenced has grown primarily through expansion on the part of commercial concerns and a desire to put to a practical test the planes' advantageous utilization. However, regardless of motives the figures stand for themselves and do show as accurate. Such demand may come from buyers who did not own planes before but later just

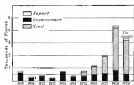


entered the aerial group, or buyers who previously owned one or more planes but who now have suggested the number. In one way or another those buyers have accepted some airplanes as a part of their group equipment. They can be called into new demand for new planes, but now only in the sense that it comes from new users by old owners or by an extension of the number of users.

The second factor does not involve a change in habits of aircraft consumption. On the contrary, it arises from the fact that when a buyer has once accepted the airplane he does not easily give it up, and thus is implied a certain stability of aircraft consumption habits.

This fact, combined with the obvious fact that planes do wear out, or for other reasons are eliminated from use as a plane grows old to an ever growing annual demand for new crafts to consume, under the economy's replacement. There is, then, in this, as in other industries, the replacement demand.

This demand does not correspond to the number of cases in which the owner of an old plane "trades in" for a new one. Neither the new demand nor the replacement



Major Distribution of Aircraft Consumption, 1925 to 1935

demand can be exaggerated and identified with particular plans coming from the manufacturers. The two types of demand exist, however, be measured with considerable precision, and through such measurement prediction of the future demand is facilitated.

Obviously during the early days of an industry practically all of the demand will be of the "new" type. This does not imply that the struggle is converting buyers to the use of aircraft is over—but it does mean that its solution has already gained such proportion and momentum that the replacement demand is now at the stage where it will begin to play an ever increasingly important part. As the number of planes in use has increased, and the planes have become older, the number consumed each year has increased and hence this type of demand has grown until it now undoubtedly accounts for fully one-third of the total demand. A statistical study, therefore, of the rate of this character becomes important to the industry.

As a further guide to the importance of the factor to the industry, aside from its stabilizing qualities on the demand side, it should be pointed out quite clearly that this replacement demand for aircraft was greater in 1934 than was the total aircraft production for 1921 and 1922. It was far greater in 1935 than was the total production in 1925 and in 1926 was equal to the total demand and output for all of 1926. For this year, 1926, the replacement demand alone will approximate more than the total annual output of the years 1924, 1925 and 1926.

Table 12.—Aircraft Facilities

Year	Land		Water		Total	
	Number	Value	Number	Value	Number	Value
1925	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1926	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1927	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1928	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1929	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1930	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1931	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1932	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1933	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1934	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000
1935	10,170	\$1,170,000	1,170	\$1,170,000	12,510	\$3,510,000

*Based on output of 1925.

The analysis of this factor of the demand for aircraft proceeds somewhat as follows: The number of planes consumed in this country every year over a definite time period are easily obtained from the authentic annual production figures plus imports and minus exports. First, now, we know the normal mortality rates, we can combine these with the past domestic sales to determine the number which should normally be eliminated in a specific period. In other words, we create a complete normal mortality curve for airplanes comparable to those for human lives and by insurance statistics. From this may be defined comprehensive expectancy tables similar to those employed by railroad companies in estimating the number of instances of seasonal accidents, telephone companies in estimating new materials and equipment needed in advance and so on down through the entire list of our intelligently managed industrial enterprises.

During the past 10 years of the industry's history there have been produced for domestic use some 12,510 planes. On Aug. 31, 1935, according to the Department of Commerce reports through licensed and identified planes, more 8,333 are accounted for. There may have been some planes in existence at this time which were not accounted for but it is reasonable to believe that this number would not exceed 10 per cent. Not a large enough figure to endanger the validity of this study. It is therefore obvious that some 4,000 planes have ceased to be used as such and have therefore been eliminated from the Nation's supply.

It should also be clear that of the approximately 4,000 planes eliminated they were not discarded in any one month or year, but rather over a considerable length of time and the elimination was more or less gradual in its process. In the absence of any official data on the sub-

Table 13.—Method of Replenishing Annual Rate of Elimination

Year	Total		Domestic		Normal		Per Cent
	Number	Value	Number	Value	Number	Value	
1925	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1926	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1927	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1928	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1929	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1930	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1931	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1932	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1933	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1934	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36
1935	12,510	\$3,510,000	1,170	\$1,170,000	1,170	\$1,170,000	9.36

*Based on output of 1925.

†Eliminated by military, total new production less total number of planes consumed.

‡Eliminated by military, total new production less total number of planes consumed.

§Eliminated by military, total new production less total number of planes consumed.

||Eliminated by military, total new production less total number of planes consumed.

¶Eliminated by military, total new production less total number of planes consumed.

|||Eliminated by military, total new production less total number of planes consumed.

||||Eliminated by military, total new production less total number of planes consumed.

|||||Eliminated by military, total new production less total number of planes consumed.

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|||||||Eliminated by military, total new production less total number of planes consumed.

|||||||Eliminated by military, total new production less total number of planes consumed.

|||||||Eliminated by military, total new production less total number of planes consumed.

THE LEHIGH



Airports Competition



Left: W. H.
Harrison,
right: A. C.
Zimmerman

By CHARLES F. McREYNOLDS
Pacific Coast Editor of Aviation

A. C. Zimmerman and William Provides for

The winner was judged largely upon the ingenuity with which the various parts of the airport were integrated in relation to each other, and for the degree of efficiency provided for in the handling of planes and passengers.

The winning design, submitted by A. C. Zimmerman and William H. Harrison, associated architects, is particularly notable for the manner in which all parts of the airport are combined, for the provision of a novel system of runways and taxiways with plane traffic controlled by colored lights and for the development of a new type of landing post, which completely removes passengers and spectators from any possible unobstructed access to the flying field, yet at the same time providing for the rapid loading and unloading of several planes at once. The idea of a star-shaped landing post with four five-

THE RECENTLY concluded Lehigh Airports Competition produced 257 designs and its object was to coordinate the efforts of architects, engineers and city planners of the United States upon the problem of preparing an air terminal plan which would avoid costly repetition of what other airports have done. General requirements were concerned chiefly with developing a land field in accordance with the following conditions: The field being proposed to be at sea level, on a level site, with an effective landing area of 3,500 ft. in all directions and with prevailing winds estimated to be of equal force in the eight cardinal and quarter points of the compass. Fixed runways were to be provided in order to handle operations in any of the eight directions, and the effective landing area was to be enclosed by a margin of 150 ft. on all sides, providing a seven to one gliding slope from the extreme limit of any runway to a point 50 ft. above the extreme edge of the margin. Lighting equipment, service and storage facilities, passenger handling arrangements, hotel and restaurant accommodations and other items important to a modern airport, were to be provided for in the design.



An artist's conception of an air view of the Quadrant airport.



Architect's drawing of the front of the passenger terminal of the Quadrant airport.

H. Harrison Wins First Prize with "Quadrant Airport" Design that Day and Night Operations in Any Wind Direction

to rise "traverse" or escape, extending out to arriving and departing planes from a central circular building which is reached through an underground tunnel from the main administration building, was originally conceived during the design of the Western Air Express terminal by S. M. Edelman and A. C. Zimmerman, associated architects. C. C. Cole, superintendent of operations for Western Air Express, is credited with much assistance by the designers in the development of this scheme, as well as assisting in the development of the hexagonal shaped service bays now in use by Western Air Express. A modification of the landing post also suggested in the winning design of the airport competition, is now in use at the Western Air Express terminal, Los Angeles, with three "traverse" radiating direct to planes from the main administration building and the entire scheme will be developed there as soon as traffic warrants.

CONSIDERING the winning airport design submitted by A. C. Zimmerman and William H. Harrison were from the aeronautical thus the architectural standpoint, we find many features of interest. It would seem that any airport designed to provide for operations in any wind direction, would of necessity be symmetrical in shape and would logically consist of a circular runway area in the center of a square field, with the terminal portions of the field devoted to airport buildings. Such a layout is shown as "A" in Fig. 1, but it is readily apparent that the airport buildings are so wisely scattered by such a plan that persons traveling out to such a field might be forced to travel half way around the post in order to reach a particular lounge or office. It is also apparent that buildings in the corners of such a field might seriously obstruct several of the runways. A simple and logical arrangement of airport areas is shown as "B" in the accompanying sketch, taking the length of one side of the field as a radius and laying off an arc from one corner of the field to the opposite corner in such a manner as to circumscribe all airport building sites in one corner of the field and at the same time produce a quadrant-shaped runway area in which can be placed runways of greater average length than in the circular area previously considered, and with all of these runways con-

spicuously unobstructed at either end. Any plane traveling from the city to such an airport, would find all buildings grouped in the corner of the field nearest to the city served, and because of this grouping an efficient relation is obtained between the various buildings and the operations conducted in them. This latter arrangement is that adopted by the first prize winner in the Lehigh Competition and is admirably illustrated by the completed plan drawing of what has been titled the "Quadrant Airport."

Within the quadrant the architects have placed eight runways, or double the number required by the rules of the competition, making it possible for planes to always operate within an average of 11 deg. of any possible wind, and at the same time for arriving and departing planes to operate from entirely different runways, greatly facilitating the handling of traffic. In developing a new airport it would be logical to start with four of the eight runways and to add the others as traffic might warrant. An important part of the present plan is the provision of three taxi ways which serve to simplify and speed the loading of planes between the landing post and runways.

In operation the runways and taxiways will be equipped with a system of colored signal lights for landing both ground and aerial plane traffic. Arriving planes will find a vertical red light at the head of every runway but one, which runway will be designated by the display of a green light at the landing runway in advance of the time. Planes taking off will leave the landing post as the dispatcher directs by the operation of the signal lights, and will travel to the head of the selected runway on a taxi strip, guided to such by the signal lights which are to be provided at every runway and taxiway intersection. Thus if a plane were taking and about to cross a runway in use at the time, it would be stopped at the intersection by the display of a red light which would change to green the moment that the runway was clear. Conversely, if traffic were being permitted to flow along a taxiway and across a runway being used for take-off, that runway would be temporarily closed by the red light at its head. In actual operation it is unlikely if any plane would ever taxi across a runway in use because the further from the depot of the two runways being

used will normally handle departing traffic, and that closest to the depot arriving traffic, permitting plane traffic on the ground to travel around the field and depot in an unobstructed line always moving in the same direction for any given work condition.

As shown by the plan of the Quadrant Airport, hangars are laid out along the semicircle which marks the division between runway area and that devoted to buildings. General traffic from the city served by the airport would arrive at the corner nearest to this city and travel directly to the administration building, or by means of a paved road along the rear of the line of hangars, to any particular hangar. Ample parking space for automobiles is provided to the left and right in front of the administration building, and also along the rear of the hangars. The entire area devoted to buildings is completely removed from the scene of flying operations and is easily adapted in location and shape to

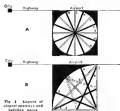


Fig. 1 Layout of airport terminal and building areas

attractive landscaping. Within this area would be placed the hotel, garage, playground and swimming pool, and whatever other facilities of this sort might be developed.

The field control tower and landing point is located in the center of a circular area toward the field from the administration building and at the mid point of the semicircular row of hangars. Even the largest land transport planes now projected would find ample room to taxi completely around the landing point to any designated landing "stand" and then on under its own power and without interfering with any of the other planes which might be landing or unloading at the same time. The distance from the administration building to the center of the landing point would be 300 ft. in the design sketched, while the individual landing stands would each extend out for 500 ft. from the center of the landing point. Entrance to the landing point vehicle is gained by means of an underground tunnel leading out from the administration building. Within this tunnel the incoming and outgoing passengers are provided with completely separate lanes, and within the landing point the lanes from plane "buses" to underground tunnel are carried off by gates or chutes, adjustable as increasing or outgoing traffic fluctuates, but always preventing any possibility of confusion in traveling to and from planes. The plane "buses" are of masonry construction for a considerable portion of their length, with a sliding end canopy telescoping into the permanent portion of the

tunnel which may be extended out to the outer door of an airplane to provide a protected passage from depot to plane entrance. This arrangement completely eliminates any contact by the passengers with dirt, wind, or noise.

If incoming cars are to be handled through the landing point a separate passage is provided in the underground tunnel, leading to a row of counting and administrative offices in the administration building. An additional lane provides for the handling of baggage and the movement of passengers between the administration and landing point.

The star shaped landing point with an underground passage to the terminal building offers several possibilities. The present plan provides a supplementary landing point on the airport level leading directly into or out of the administration building. This supplementary landing point for a private owner to land and load to the nearest building, go in and buy a ticket, and go on as though the underground passage to explore upon a regular transport plane for some distant point. On this supplementary landing point might be used for taxiing and right steering operations in a system of looping and traffic operation from the transport operations. A modification of this star shaped landing point is now in use by Warner Air Express. Three tunnels run out to the planes on the field level directly from the administration building. By slightly increasing the radius from the center of the landing point out to the planes it is possible to provide six tunnels instead of five, and that handle new planes simultaneously around one landing point. A further logical development is to establish two or three landing points in conjunction with the one administration building, with separate underground passages out to each point, thereby making it possible to handle as many as 20 planes simultaneously. This is graphically illustrated in Fig. 11, A, and it has been suggested by the architects that where a group of three landing points were employed, the central one would be placed at the greater distance from the administration building, providing the dispatcher, who would be located in the central unit, with a better command of the field traffic. A further possibility is to segregate traffic among the three units, handling all arriving transport planes at one point, all departing transport planes at a second point and receiving and mail and all baggage and private plane traffic. Of course the dispatcher would use his facilities as experience and current traffic might dictate.

There is a landing point as proposed in the winning design of the Lehigh Airports Corporation, providing three story arrangement with passenger handling on the ground floor, pilot's room and registration office, weather office, and radio room on the second floor, and the control room, occupied by the dispatcher, on the third floor. From his position the dispatcher can at the same time see every plane that is taking off as well as the entire field area. There are no airport buildings opposite the ends of any of the runways except in the case of the runway which bisects the quadrant landing area and which runs directly toward the landing point. With 15 ft. of the possible 150-ft. of direction completely unobstructed, the airport would be laid out with relation to the known winds so that the runway in question would never be used for take-off. This one runway could also be used to pass on either side of the landing point, completely removing any actual hazard which might occasionally arise in the present arrangement.

Plans provided for in the winning design are of



Fig. 11. The manner in which the star landing system can be progressively developed to handle from three to twenty planes simultaneously and efficiently.

the rectangular type with sliding doors at each end and a maximum door width of 125 ft. to assure the necessary condition of any land plane now projected. As required by the conditions of the contest four-fifths of the hangar space is devoted to storage and one-fifth to service. All hangars are set with door openings at right angles to the quadrant are entered off facing the field, insuring greater privacy to the various companies whose hangars and also giving each hangar an adjacent open space which can be used without cluttering up the main taxi apron extending all along the row of hangars on the field side. Facing windows are located on the main taxi-apron and an angle area is set aside just off the apron along the border of the flying area to left and right of the landing point, for the purpose of parking planes. One of the most commendable phases of this arrangement of administration building, landing point, hangars, taxi apron and plane parking area, is that it concentrates all of these activities in the most efficient way, greatly reducing the confusion which might otherwise arise in contact between the general public and the various air terminal activities.

The administration building is of modern design, being appropriate to such a modern problem as an airport. The main portion of the building being occupied by the ticket office and waiting room, with rest and rest rooms in the basement, and the ceiling of the main waiting room extended two stories as in most rail depots. Three wings, one story in height, extended to left and right, and toward the field from the main portion of the building. They are occupied by the restaurant, administrative offices, association offices, freight and mail handling rooms, etc. (On the roof of each of these wings are open air observation platforms is provided for the use of the general public in observing field operations.) An additional ground level area is open to the public to left and right of the administration building and adjacent to the circular taxiway around the landing point but separated from it by appropriate fencing. Casual stables and enclosures are provided for in this area.

Because the competition program insisted itself only with the problem of passenger terminals, no particular attention was given in the winning plan to the handling of large volumes of aerial freight and express. Although it is true that almost all railways handle their freight shipments at separate terminals, from their sea-

sonage facilities, it is the opinion of Mr. Zimmerman, that air freight and express shipments will in general be handled from the new field as passenger traffic, for a time at least, the freight depot itself to be set apart from the passenger terminal. This freight depot would probably consist of a long two way shed similar to those used in most rail freight terminals, permitting the arriving planes to range along one side of the shed under a protecting roof, and to discharge freight, while the departing planes would move along the opposite side of the shed taking on their freight.

This airport as laid out should meet all passenger air transport requirements for several years to come in the opinion of the designing architects. It would be placed at a natural distance from the city served, relying upon good surface arrangements to speed a natural connection to downtown area. If rail bus air fields can be developed they will work in admirably, it is thought, with the major airport plan, since passengers arriving at the main terminal could there replace in smaller two places and be flown to their various rail bus destinations, which will in any case probably be too scattered to make worth while to move the major airports much closer to the centers of our cities.

As provided by the requirements of the competition, all runways and buildings, where practicable, are designed to comply Portland request. The design of all the buildings connected with the contest, is most attractive and carries out the futuristic motif so appropriate to an airport, but of course the general type of architectural

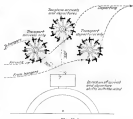


Fig. 11-B

treatment of any or all of the airport buildings might be allowed at will without affecting in any way the present plan of field layout.

If there are to be airports of the future developed in sparsely settled portions of the country and surrounded by country residences from which the workers would daily go to their employment in the large business centers, such a development is not very much in the future and has not been restricted by the designers of the plan awarded the prize by the judges of the Lehigh Airports competition. This design is not particularly worthy of study because it is a design for the present planning to solve the problems of the present and forecast future similar as passenger air transport terminal traffic handling in concerned.

AVIATION PROGRESS UNDER THE Soviet

A description of the work of the Russian Central Dynamic Institute, with some of its practical

Aero-Hydro-

By V. M. PETLIAKOV

THE PURPOSE of this article is to describe the work and the product of one of the scientific institutes of the Soviet Union, the Central Aero-Hydro-Dynamic Institute (TsAGI), which is responsible for the designing and building of the plane, "The Land of the Soviets," used in the 12,900-mile Moscow-New York flight, as well as of many other Soviet planes.

The "Land of the Soviets" is of the Model ANV, named after the well-known Soviet engineer, Andrei Nikolayevich Tupolev, who designed also "The Wings of the Soviet," which recently completed a flight covering many European countries, and other Soviet planes with long-distance records to their credit.

The "Land of the Soviets" was built in its entirety in the Soviet Union—with the exception of the engines of the German BMW type, and the wheels. The author of this article has had the pleasure of working with Mr. Tupolev as one of his closest associates in the construction of the plane. It is necessary, perhaps, to make it clear that the plane was not only constructed in the Soviet Union, but also solely of native materials (apart with the exception of the engines), and by Soviet workers.

The Tsagi Institute was founded in 1918; in that difficult period during which the country was torn by civil war and foreign intervention. The founder of the Institute was Prof. N. S. Zhukovskiy, whose name as the founder of Russian theoretical aviation research is well-known abroad and in the U.S.S.R. The very name of the institute signifies that the objects of its research work are two elements—air and water. The chief work of the institute consists in studying the mechanical powers of air and water in order to utilize technically their useful properties, and to apply them for economic purposes. On the other hand, the Tsagi Institute also studies the adverse characteristics of air and water in order to overcome them. The purpose of its work, then, lies not only in the field of aviation, but is considerably wider, including aviation as only a part of its work.

The basic principle on which the work of the Tsagi Institute has been built is that of carrying out each piece of research to a concrete result, in order to permit its immediate application by industry. With regard to aviation, for instance, the final purpose of its work is the creation of airplane models which can later be produced in the required quantities in Soviet factories. In order to

V. M. Petliakov is assistant to A. N. Tupolev, designer of the "Land of the Soviets" and other Russian planes. He describes the extensive equipment for aviation research which is available to Soviet engineers, and the way in which research is coordinated to fit it to the practical problems of airplane construction. From the emphasis placed upon laboratory work, he proceeds to detailed descriptions of "The Land of the Soviets" and other Russian planes, together with some of their performance figures on recent notable flights.

achieve concrete results it is necessary to go through considerable preliminary research work. This work is along two channels, theoretical and experimental.

THE TSAGI INSTITUTE has a number of divisions. The listing of these might possibly be considered as a considerable preliminary research work. This work is

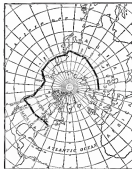
1. Theoretical Department. This division has provided us with considerable material concerning the dynamics of the flight of airplanes, the theory of the method of calculating propellers, and the methods of calculating the performance of the plane.

2. Experimental Aerodynamics Department. The principal work of this department is connected with studies in regard to conditions of aerodynamic tunnels. This department has a large wind channel, 50 meters long (164 ft.), with two working cross sections of three and six meters (9.8 and 19.7 ft.) in diameter. The cross section of the wind channel is polygonal. The tunnel gradually tapers from the larger to the smaller diameter. It is of the closed circuit type, and is built as two separate tunnels, one of three and one of six

meters in diameter, so that when the six-meter tunnel is required for work the connecting part can be removed, and work can be conducted in the shorter six-meter tunnel. [This is a very novel feature in wind tunnel design. The tunnel consists with a tunnel which is to be operated at two different diameters is to fit a throttling valve into the throat, to reduce the diameter and thus shorten the experimental series by removing the entrance and exit cones into it. The method used at the Tsagi Institute of traversing the entrance section of the wind tunnel body is shown in one of the illustrations in this article.—ED.] The wind tunnel and most of the measuring instruments have been designed by members of the Institute.

The tunnel gives a very good efficiency, which is shown by the fact that the velocity of the flow can be maintained at 100 meters per second (223 mph) through the use of a 6-meter (19.8 ft.) propeller with a motor of 630 hp. Until recently this tunnel was the largest in the world. Last year a larger tunnel—55 ft. in diameter, that is only 4 in. larger than the Tsagi tunnel—was constructed at Langley Field in the United States.

3. Engine Department. This department, aside from research work in the construction of other types, also



Route of the "Land of the Soviets" to New York from Moscow in New York

conducts work in the experimental construction of aviation engines, which later on are put into actual operation. The work in testing engines on platforms furnishes considerable material for use in selecting types for different airplanes. This work also helps to solve the problems of installation and operation of engines in airplanes. The installation of a special low-pressure chamber was completed recently, and a study of the performance of engines at high altitudes is being started.

4. Department for the Testing of Aviation Materials and Designs. The work of this department is to specify the technical conditions which must be met by all materials that are used in the construction of planes; to create new methods of the testing materials do not



The "Land of the Soviets" at Washington, D.C., after its flight from Moscow

and while the plane is in flight. At the same time good working conditions for the propellers are obtained by long drive shafts projecting in front of the wing. These shafts are enclosed in a streamline housing fixed onto the leading edge of the wing.

The data the plane has made were more than twenty flights, as long as one hour. When the writer visited the Dessau works, a number of minor changes were being made. Because of the many radical features in its design and because all tests were not completed, the construction was not so reliable about saving very much as to their expenditure and, as well, about any of the details of its construction.

At first two L-55, 500-650 hp. and two L-8, 400-hp. engines were used though the L-55's will be replaced by two Junkers L-88 800 hp. engines and thus give a total of 2,800 hp. The weight of the G-38 engine is stated to be between 26,000 lb. and 30,000 lb. (13 to 14 metric tons), while the gross weight is from 14,300 lb. to 54,000 lb.



Close-up of air shaft of Junkers showing Division G-38's mounting structure and shaft structure

(30 to 24) metric tons). Thus the useful load is between 11,200 lb. and 26,500 lb. (6 to 11) metric tons). It is stated that with 4,500 lb. pay load the range is about 2,200 miles.

The performance has not yet been determined though the high speed is estimated at about 125 mph.

The airplane was designed to use the Junkers heavy of Diesel engines which have been undergoing flight tests in Dessau in a small single engine Junkers monoplane. With the Diesel engines it is stated that the per-

formance of the G-38 will be greatly increased allowing a larger carrying capacity and an increased flying range. The Diesel engines were not installed for the first flights as it was thought undesirable to have too many radical and new details under test at the same time.

The G-38, at the present time has its cabin empty though the plane was laid out as a long distance freight and passenger carrier. Normally it should carry about 30 day passengers with the rest of the load being freight. If necessary, with a shorter range, 48 passengers and baggage can be carried though the cabin does not appear as though it would be very comfortable when housing so many persons. As a night passenger transport, beds can be provided for 16-22 people.

In 1930 Professor Hugo Junkers obtained a patent on the principle of a flying wing containing passengers, crew, engines and all equipment and load, thus eliminating the usual parasite resistance of wings and engines.

The G-38 approaches this in that its four engines are completely contained in the wings while the fuselage has been reduced and shortened until it becomes merely a large cabin extending into the stubs of the deep wings and tapering in cross section very rapidly into a support for the highest tail surfaces.

This machine was designed as a successor of the Junkers G-100, a projected 100 passenger multi-engine design containing all passengers, crew and engines in the wing and differing from the G-38 chiefly in that the G-100 has no fuselage and its stubs had two short struts projecting in front of the wing to support the stabilizer in front of the mainwing (as in the "Ehrh" or Dack type). The vertical control surfaces are mounted on the wing tips and thus reduce all resistance to a minimum.

However, in the design and construction of the G-38, more conventional methods of flight control were followed though the place contains most of the essential elements to characterize it as an intermediate step toward the development of the flying wing. The G-38 wing was made sufficiently deep that a great portion of the load would be carried in no interior and thus give a more efficient type of structure by having the load and the supporting members close together. In addition the resistance is reduced as the only disturbing units producing parasite drag are the landing gear, control surfaces and the comparatively small fuselage.

The aspect ratio of seven is quite good for a plane of these proportions. The wing has a great taper providing a very large chord and thick section at the root. It is interesting that the wing is almost flat on top and that the taper produced what might be termed an effective dihedral.

Also the wing has an effective sweep back by having the trailing edge be almost completely on a straight line connecting the trailing edges of the wing tips and having the taper in plan form close the leading edge and consequently the median line of the wing to nearly horizontal. The wing chord at the root extends almost the entire length of the stub portion of the fuselage making for good longitudinal stability with varying wing loads and thus eliminating the tail tank type of balance found in the tail of many European transport planes.

The wing is very deep, being about the same as the depth of the fuselage. The upper surface of the wing is flush with the roof of the cabin and the lower surface is at a level with the floor of the cabin. Under the cabin floor is another compartment used to house baggage and freight and at the same time act as a "crash space." In



View of a portion of the roof of the G-38 showing the leading edge structure during the early test flights.

the event of a crash this compartment would be destroyed and the force would be absorbed in deforming its structure, thus protecting the cabin. It is stated that in all Junkers designs there is considerable structural material under the cabin to provide protection in the event of a crash.

Usually the wing structure extends under the cabin but in this plane, because of the great depth of the wing this was not possible and therefore this compartment was added. This portion under the cabin gives the G-38 the appearance of a high wing monoplane. In German it is called a "Hochle Dedler."

The structure of the fuselage is of open section diamond shape riveted into the form of a deep truss and reinforced by an outside skin of corrugated duralumin. However, the rounded nose of the fuselage and the leading edge of the lower portion of the wing, as well as the engine drive shaft bearings are covered with special flat duralumin sheet. Though this might be to facilitate the manufacture of sheet surfaces with bends around two radii, it is interesting to know that there are being completed at the Junkers plant two identical high speed two motor prototypes, one of which is covered with flat sheet and the other with corrugated sheet and is investigating the performance of the two designs.



Direct view of the tail end of the plane showing the cabin and motor in the group of work in the photograph

As mentioned above the interior of the cabin is as yet unfinished. However, it is known that the cabin would lead easily into accommodation of freight forward where there can be as windows because of the deep wings at the sides. Freight can be carried in that portion which extends some distance into the wing roots. Behind this the cabin is narrower and has some windows at the sides above the wing. Windows will be added forward later. The interior can easily be divided into various compartments either for sleeping passengers, day passengers and freight, private compartments, etc. In addition, two lavatories and washrooms are provided.

Construction with the wing spans the length of the fuselage is short, giving a close cockpit seat. In front of the wing is a small projection or nose containing a large window compartment for the operating officers. Behind and above this compartment is the cockpit for the two pilots sitting side by side with dual control. The cockpit is very large and roomy with glass covering and the visibility is quite good considering the great size of the machine. The windows may be opened should the pilot wish a better view. In addition, there was mounted temporarily a mechanical "feeler" under the nose to indicate proximity to the ground when landing.

The plane is actually steered by the officer in the nose and the pilot fulfills a role comparable to the pilot of a seaplane. Behind the plane's cockpit is a large vertical instrument board on which are mounted all of the instruments and controls for the power plants. A chief mechanic is stationed at this panel and thus relieves the pilot of the task of operating the engines.

At the sides of the engine control compartment are three windows looking to the rear divided into two compartments in the wing. However between the fuselage and the wheel engines, and accessible from these openings,

are small components in the leading edge of the wing or radio operator's shields for two pilots who could sit unobstructed and look through the windows in the leading edge.

These components, as well as those for the engines, located symmetrically but oriented backward, as well as the components in the center of the wing for tanks and fuel gauges, are all distributed between the main structural members of the wing trailing. However, the trailing is laid out so that when an airfoil in the air foot deep wing structure is no more noticeable than that in the fuselage center of a well-designed transport plane.

The wing is built entirely of duralumin with some steel parts where high stresses occur. In detail, it follows usual Junkers practice except at points where stresses were made by stresses generated by the great size. Like other Junkers designs, the coverage is of corrugated duralumin nested to a sandwich truss structure visible slightly to permit the components in the wing. The space in the fuselage projects into the first form of the wing truss increasing the use of the skin and reducing some of the loading on the wing structure.

One of the most interesting details is the power plant installation. The engines are water cooled and have their radiators projecting below the lower surface of the wing. The radiators are each individually retractable into the component above to regulate the cooling or to facilitate repairs in flight. However, it was found that these radiators, as well as the engine exhaust piping, loaded the engine components so much that a special supporting system is being installed. The engine components are quite heavy, allowing one to walk around comfortably. The writer was permitted to enter the in-



board engine rooms when there were about four men already there. This is a great advantage for the repair of the engine without removing them and also for facilitating inspection during flight by both mechanics and the pilot's affairs.

The Junkers J-88 gas turbine engines are rated at a maximum of 800 hp at 1550 r.p.m. and normal 700 hp at 1800 r.p.m. The propellers rotate at a slower speed than the crankshaft, being fitted with a reduction gear having a ratio of 1:1.5. These engines are the active cylinder 60 deg Vee type with four valves per cylinder. The bore and stroke are 6.3 in. by 7.8 in. (160 mm by 195 mm). The cylinders are individual with each having a separate water supply for cooling the cylinder as well as the valve case and valve guides. The valves are driven by overhead rods with no rocker arms. The rods are driven by

vertical shafts. With normal gear gear for propeller drive, the weight is 1435 lb. (650 kg.) at 187 hp per hp. The two reduced engines are similar to the J-88 engine except that they are instead of twelve cylinders and have half the power.

Later, the Junkers Diesel engine will be substituted. These engines are the result of a long research program that is just about completed. At present the six cylinder Junkers Diesel engine is rated 700-800 hp and weighs 2245 lb. (1020 kg.) at about 3 lb./hp. One of the most important features in the development of this engine is the water cooling. The system provides for four radiators which completely surround the fuel in 8000 sec. and mix it completely with the combustion air by means of the proper turbulence and rotation that is controlled in such a manner as to have proper air flow and gas practically perfect combustion.

Another important feature is the turbo-driving pump with a running speed operating at a peripheral speed of 658 ft per sec. (200 m/sec.). This pump also has a



Also: The Junkers J-470 gas turbine engine of which one used in the G-38. Also, one of the J-470 engines of 400 hp, which has an overall length of 10 ft 6 in., a height of 4 ft 6 in., and a weight of 1100 lb. Two of these are used in the G-38.

special shaft protected from vibration in the drive shaft. Special attention had to be paid in the design of the cooling pipes, due to the resistance at these high speeds.

A great deal of study was devoted to the crankshaft bearings. Traditionally every conceivable type of bearing was tried and it is not clear if this point has been completely solved as yet. It has been said that many successful experiments have been carried out on torsional vibrations in this powerplant and that an entire new system employing a double crankshaft has been developed. The crankshaft is interesting in that it also forms the engine mount. It is stiff in all directions, being a casting that must be exceptionally difficult to produce.

When in the engine works the writer saw some parts for twelve cylinder Diesel engines. These were of the horizontally opposed design with provision that could be mounted in either vertical or horizontal. The horizontally opposed cylinders have integral handle making an exhaust for 12 cylinders. There are no valves or crank and, as they are all injection engines, there is no distributor or injection system. Therefore it is obvious

that many of the usual causes for engine failures have been eliminated. Junkers is in production on similar Diesel engines for motor trucks and marine use as well as for stationary engines. Both gasoline and Diesel engine development at Junkers are under the direction of Dr. Mader and Dr. Gutzmer.

It is planned to use only one compressor for all four Diesel engines for the G-38 and similarly only one starter compressor. At the present time there is a small number of driven air compressors behind the left inboard engine. This is used to force an under pressure in the present gasoline engines and to start them by a starting slide valve mounted on each cylinder and driven by the crankshaft.

At the front of the gasoline engines are the spur type reduction gears coupled to an oil clutch developed by Junkers. Its purpose is to reduce vibration between the propeller and engine. The long propeller shafts prevent well out in front of the leading edge where they drive four blade wooden propellers fitted with spinners forcing into the streamlines over the skin.

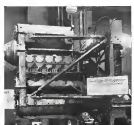
The control surfaces of the plane are especially interesting. They are so laid out that no exceptional forces are required by the pilot to maneuver the craft. All of the movable control surfaces consist of separate units not fixed to the fixed surface in front as is usual practice, but rather having a definite slot between the surfaces. The ailerons are auxiliary airfoils mounted below the main wing with the aileron leading edge below and well forward of the trailing edge of the wing. In this way the Junkers ailerons differ from the wing ailerons and fins developed in America. In the writer's understanding that the wing auxiliary aileron is located entirely behind the trailing edge of the main wing. However, the Junkers people, as well as Mr. Wright, both claim that such a device will definitely increase the lift and then lower loading and increase speed as well as lower the fuel and increase the high speed. This device was tried to a small Junkers monoplane a few years ago. Considering that the wing and power loadings of the machine are normal for a plane of this size (14.15-17.3 lb. per sq ft. and 18.38-22.5 lb./hp.) and that the wing is proportionally larger than the load speed should be quite low. When one considers that all of the power was contained in calculating the above power loading, it will be interesting to note the performance with only three engines running.

As for the horizontal control surfaces are balanced in a similar manner. The hinges are set back from the leading edge and are supported from above, thus affording some aerodynamic balance. In addition, small weighted arms are mounted on each end of each control surface and, as these project well forward of the control hinges, the center of gravity of the control surface is moved forward.

The tail is of biplane design to make it more compact and thus increase its structural efficiency. There are three vertical surfaces, a fixed fin and rudder in the center above the fuselage and two vertical rudders mounted at the extreme wing tips of the horizontal surfaces. There is only one fixed vertical surface, that is in the middle, the outboard rudders being no less in front of them. The upper stabilizer is adjustable and supported by the middle vertical surface and by two inverted Vee struts, one in front of each outboard rudder. These inverted Vee struts have the apex of the Vee over the leading edge of the upper stabilizer and their lower ends attached

to the lower stabilizer. Like the ailerons, the elevators have weighted arms for balance.

The landing gear is especially interesting. It consists of five wheels, a tail wheel and two main wheels. The wheels are located in the usual positions below the wings. The wheels are all Electric (magnesium alloy) castings and all five are fitted with Kearsy pneumatic brakes, similar to those made by the same firm for use on railway cars. The tandem wheels are coupled together on each side by electric linkages acting as a lever with a wheel at each end and a fulcrum in the middle between the



The Junkers double winged cylinder Diesel engine rated at 800 hp.

wheels. The fulcrum is attached to a vertical tube which transmits the landing loads to the wing through a shock absorber of rubber chord.

The Elevator lever is hinged so that the wheels are free to rock about the fulcrum and thus adjust themselves to the unevenness of the ground. As the shock absorber collapses, the landing gear returns about the lower fulcrum of the fuselage to which are connected three large landing gear support tubes. The tubes are fitted with ball and socket joints so that the rocking of the wheels does not affect the seats or the shock absorbers. To keep the wheels approximately the same level when the plane is in the air there are short strands of shock absorber chord stretched between the axle and the main shock absorber. Later the wheels will be joined by a single towing baring the frontal area of one wheel.

Some of the general data on the G-38 may be summarized as follows:

Span (46 m.)	147.3 ft.
Length (35 m.)	115.5 ft.
Height (15.5 m.)	18 ft.
Wing area (290 sq m.)	3,120 sq ft.
Aspect ratio	7
Maximum wing thickness (218 mm.)	8 1/2 in. (19 in.)
Weight empty (24,000 kg.)	28,500-30,800 lb.
Loaded load (6,611 kg.)	14,500-24,500 lb.
Gross weight (25,249 kg.)	44,200-54,000 lb.
Rated power	2,400 hp
Wing loading	14.15-17.3 lb./sq ft.
Power loading	18.38-22.5 lb./hp

THE BUYER'S LOG BOOK



AC Mica Spark Plug

IN ADDITION to the present line of AC plugs a complete line of mica insulator spark plugs especially designed for aircraft engines, is now offered by the AC Spark Plug Company, Flushing, N.Y. These have been graded according to heat range, making it possible to produce the correct plug for extreme operating conditions.

The plug is made from light by an electrical process and the side wire is electrically welded, preventing all water greater electrical and three times greater heat conductivity, according to the manufacturer. Terminal connection and center wire are fused together electrically, eliminating danger of the connection becoming loose from vibration. Long insulation is provided between terminal and shield, preventing spark flashover at high altitudes. The plugs are flameproof with metal parts designed to prevent rattling.

Ball Bearing Live Center

THE ball bearing equipped tailstock live center, lately placed on the market by the Bradley Company of Bridgeport, Connecticut, is designed to meet the greatly increased speeds at which metal removing machines now operate. By forming a rigid and non-deflecting support for the work, this design aims to reduce chatter and slowness of cutting. The universal contact bearing has solid, high carbon chrome steel outer and inner cones and is so constructed that the cones, when exposed to a deficiency produced condition.

When the center is set up, the work is brought close to the tailstock proper so that the inherent rigidity of the bearing is not offset by deflection of either parts which would result where there is considerable overhang. Another feature in the design is the bearing clearance or fit and location on the side of the work and protrusion from the direct force of the cutting compound. The relative bearing helps to keep the inherent fit and gives considerable end-chase away from the bearing seat. The ball bearing is so mounted

in the center that the outer cone rotates with the nose of the center and the nose, as point, is ground after assembly. Since the center also rotates with the work, regardless of the nose in infinitely regulated feed may be accomplished without stopping the center from the machine. The bearing is now manufactured in the most frequently required speed sizes, but types of any type or size can be made to order.

Cable Splicing Device

A new Universal Cable Splicing Device has been placed in production by the National Steel Products Company, Dayton, Ohio, a subsidiary of John S. Galt Manufacturing Company. This tool provides for holding the



Universal cable splicing device

double and triple in position while the wires are being made by the universal action of the jaws which tighten the cable around the dielectric. It is adaptable for any size flexible and aluminum cable and can be reversed allowing an even pull on each strand during the splicing operation.

Silent Chain Bench Lathe

A new silent chain power drive adapted to a complete line of bench lathes has just been announced by the South Bend Lathe Works, South Bend, Ind. The drive is available on both the 9 in. regular and 9 in. junior models. An interesting feature is that the entire endshaft and synchronously idler shaft drive, and the pulley belt cone pulley are mounted over the lathe headstock, out of the way of the operator, chips and dirt on a substantially noiseless bracket and lifting table which allows easy shifting. The adjustment for the belt stretch is independent of the sliding table.

The new model 9 in. bench lathes are built around new cutting universal precision lathes and for general finishing, converting tool rubbing, amassing, shift and accumulator turning, brooding, grinding, cutting and a large per cent of other automatic replacement and repair machine work.

TRADE CATALOGS

THE COMPRESSOR, "The Engineer Air Compressor Manual" is the title of a 45 page booklet recently published by the Browne Manufacturing Company for the use of aircraft mechanics and operators. A catalog of the improved Browne line is included.

BOOKS & 20. A revised 15 page catalog published by the Holsen Aviation Corporation describes the important features of the 5-28 with special attention to the interior. Specifications and performance figures are given. Illustrations of the plane over land and in water are included together with views of the plane.

TURNING CENTER. The Penn-Cable Machine Company is now publishing a condition, informative booklet and catalogue entitled "Turning With Tungsten Carbide." The first part of the booklet is devoted to information on the Tungsten Carbide, while the latter portion is the catalogue on their new Carbide-Lathe.

IN ALUMINUM. A folder describing specifications of the Le Blond "60" has just been revised by the Le Blond Aircraft Engine Corporation. Illustrations of the main parts and of airplanes with Le Blond installations are shown in the folder.

Gun Steel Propeller

THE NEW Gun steel propeller designed for 30 hp engines is now being manufactured by the American Specialties Company, San Marcos, Calif. The blades, which are interchangeable, measure 4 ft. 1 in. in length, 50 in. at the greatest width and 4 in. in thickness, being constructed to preserve the life of each bolt. The leading edge of the blade is positively straight with an adjust of 6 in., while the trailing edge is beveled and tapered from the greater width toward the tip of the blade and toward the hub.

The hub is constructed of steel, hardened and heat treated. The hub arm which holds the adjustable blades is 12 in. long, 2 1/2 in. wide and 1/2 in. thick. The portion of the hub holding the blades is 10 in. long, 2 1/2 in. wide and 1/2 in. thick. Arms are of chrome vanadium steel bolt treated and oval design, measuring 1/2 in. thick and 3 in. wide. They are fastened to the hub and blades by nickel steel bolts 1/2 in. x 1 in. held in position by cotter pins and lock washers, and are of SAE rating. The total weight of the propeller including spinner is 55 lb.



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Average net paid A.B.C. in 1934: \$10,000 copies

Total circulation February 15 issue, 34,000 copies.

overhauling etc. Representatives throughout the industry have turned up their sales promotion plans to make use of all types of sales producing tools.

The majority of them will "give the gas" to their spring selling drive by exhibiting at the outstanding show of 1935—The International Aircraft Exposition, St. Louis, February 15 to 23. It will be those that

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